

SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution) Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai Accredited by NAAC-UGC with 'A++' Grade (Cycle III) & Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech. IT) COIMBATORE-641 035, TAMIL NADU



DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name	:	Dr.A.Arun Negemiya, ASP/ Aero	Academic Year	:	2024-2025 (Even)
Year & Branch	:	III AEROSPACE	Semester	:	VI
Course	:	19ASB304 - Computation	al Fluid Dynamics f	or A	erospace Application

UNIT V – FLOW FIELD ANALYSIS AND TURBULENCE MODELS

Case Study: Stage Separation Aerodynamics for Space Systems

Stage separation in space systems involves separating rocket stages, requiring careful aerodynamic design and analysis to ensure a safe and successful separation. This case study will explore the aerodynamic challenges and solutions during stage separation, including interstage interference, dynamic stability, and separation mechanisms.

Challenges of Stage Separation Aerodynamics:

• Interstage Interference:

During separation, the flow field around the separating stages can create complex and potentially destabilizing aerodynamic forces and moments.

• Dynamic Stability:

The dynamic stability of the vehicle during separation is crucial for ensuring a safe separation and preventing unwanted motions or collisions.

• Uncertainty in Flight Conditions:

Variations in flight path angle, altitude, and stage separation timing can significantly impact the aerodynamic forces experienced during separation.

Solutions and Approaches:

• Aerodynamic Separation:

In some cases, aerodynamic surfaces (like elevons) can be used to actively control the separation process and ensure a safe separation.

• Separation Motors:

For more complex separation scenarios, separation motors can provide the necessary thrust to separate the stages.

• Simulation and Analysis:

Computational Fluid Dynamics (CFD) simulations and 6-degree-of-freedom (6DOF) analyses are used to model the separation process and assess the aerodynamic forces and stability.

• Wind Tunnel Testing:

Wind tunnel tests can provide valuable data on the aerodynamic characteristics of the separating stages and the effects of interstage interference.

• Multi-body Dynamics:

Modeling the separation process as a multi-body dynamics problem allows for a more comprehensive analysis of the interactions between the separating stages.

• Separation Mechanism Design:

The design of the separation mechanism (e.g., explosive bolts, and clamp bands) is crucial for ensuring a reliable and safe separation.

Case Studies and Examples:

• Ares I Launch Vehicle:

NASA's Ares I launch vehicle involved powered stage separation, requiring timeaccurate simulations of the separating stages under the influence of time-varying thrusts from separation motors.

• Space Launch System (SLS) Booster Separation:

The separation of the SLS boosters is a complex multi-body problem that requires sophisticated simulation techniques to account for the interactions between the separating stages.

• Parallel-Staged Shuttle:

Studies on parallel-staged shuttle separation have investigated the feasibility of separating the orbiter from the booster at various points along its ascent trajectory.

• Hypersonic Vehicles:

Numerical analysis of hypersonic flow past parallel-staged vehicles during stage separation is performed to analyze the aerodynamic characteristics and dynamic stability.