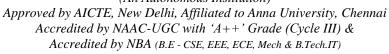


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COIMBATORE-641 035, TAMIL NADU

DEPARTMENT OF AEROSPACE ENGINEERING

Dr.M.Subramanian, Faculty Name

Academic Year

2024-2025 (Odd)

Year & Branch

Prof & Head/ Aerospace

Semester

III Aerospace

Course

19ASB302 – Finite Element Method for Aerospace

Unit: 1

A Cantilover beam of length L is leaded with a point at free end. Find the maximum deflection and bending moment using Rayleigh-Ritz method using the function y= A 1- cos(Toc)

(i) To find maximum deflection:

The deflection for the above beam is y= A /1- cos Tex

The boundary anditions for this beam are at x=0, the deflection y=0 and The Slope dy/d= 0

From eqn (), y cat x=0) = A { 1- cose } =0

Also Eqn (1) implies $\frac{dy}{dx} = A \left\{ 0 + \frac{\pi}{2L} \sin \frac{\pi}{2L} \right\}$

At x=0, dy/da = A for TI sine = 0

Sinetho boundary condition are satisfied by the trial deflection fun correct trial functions. function, it is to

T,P,E -> TT = U-W





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Prof & Head/ Aerospace Year & Branch **III Aerospace** Semester

Course 19ASB302 – Finite Element Method for Aerospace

Unit: 1

$$U = \underbrace{EI}_{2} \int_{0}^{\infty} \left(\frac{d^{2}y}{dx^{2}}\right)^{2} dx$$
To find the value of strain energy, let us differentiate the deflection function the times deflection function the stimes
$$\frac{dy}{dx} = A \int_{0}^{\infty} 0 - \left[-\sin \frac{\pi x}{2L}\right] \frac{\pi}{2L} = \frac{A\pi}{2L} \sin \frac{\pi x}{2L}$$

$$\frac{d^{2}y}{dx^{2}} = \frac{A\pi}{2L} \left(\cos \frac{\pi x}{2L}\right) \frac{\pi}{2L} = \frac{A\pi}{4L^{2}} \cos \frac{\pi x}{2L}$$
Now
$$\left(\frac{d^{2}y}{dx^{2}}\right)^{2} = \left(\frac{A\pi^{2}}{4L^{2}} \cos \frac{\pi x}{2L}\right) = \frac{A^{2}\pi^{4}}{16L^{4}} \cos \frac{\pi x}{2L}$$

$$= \frac{A^{2}\pi^{4}}{16L^{4}} \left[1 + \cos \frac{\pi x}{2L}\right] = \frac{A^{2}\pi^{4}}{16L^{4}} \cos \frac{\pi x}{2L}$$
The strain energy
$$U = \underbrace{EI}_{2} \int_{16L^{4}}^{2} \left(\frac{1}{4} \cos \frac{\pi x}{2L}\right) dx$$

$$= \underbrace{EIA^{2}\pi^{4}}_{64L^{4}} \left[\frac{1}{4} \left(1 + \cos \frac{\pi x}{2L}\right)\right] dx$$

$$= \underbrace{EIA^{2}\pi^{4}}_{64L^{4}} \left[\frac{1}{4} \cos \frac{\pi x}{2L}\right] dx$$

$$= \underbrace{EIA^{2}\pi^{4}}_{64L^{4}} \left[\frac{1}{4} \cos \frac{\pi x}{2L}\right] dx$$

= $\frac{EIA^2\pi^4}{64 L^4}$ (L-0) + $\frac{L}{\pi}$ (SiGH-SiNO) = $\frac{EIA^2\pi^4}{64 L^4}$ (L-0) + $\frac{L}{\pi}$ (0-0) = $\frac{EIA^2\pi^4}{64 L^3}$ Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering





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Proof done H-P. your = Py at x=1

=
$$P\left[A\left(1-\cos\frac{\pi}{2L}\right)\right]$$
 at x=1

= $P\left[A\left(1-\cos\frac{\pi}{2L}\right)\right]$ = PA --- G : $\cos\frac{\pi}{2}$ = O

Substituting the Values of equation G G G in equation O , we get,

 $T = U - W = \frac{EIA^2 - TI^4}{64 L^3} - PA$

For mainmum potential energy andition, $OTI = O$
 OA
 O

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