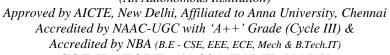


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COIMBATORE-641 035, TAMIL NADU

DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name

Dr.M.Subramanian,

Academic Year

2024-2025 (Odd)

Year & Branch

Prof & Head/ Aerospace
III Aerospace

Semester

 ${f V}$

Course

19ASB302 – Finite Element Method for Aerospace

Unit: 1

* Analyso a simply supported beam Subjected to a uniformly distributed load Horought using Rayleigh Ritz mottood. Adopt one parameter trigonometer a function. Evaluate the maximum deflection and BM compare with the exact soluction. A promoner B (AU, 2008) 2=0 From given condition one perameter trigonometer is given by

y = a1 Sinta > a1 is Ritz perameter. we know that Total potential energy of the beam, TT=U-H Where, U= Strain energy H = Work done by external force The Strain energy, U, of the beam due to bending is given by, $U = \frac{EI}{2} \int \left[\frac{d^2y}{dx} \right]^2 dx - 3$ $\frac{dy}{dx} = a_1 \cos \frac{\pi}{1} \times \frac{\pi}{L}$ $\frac{d^2y}{dx^2} = -\frac{a_1\pi}{l} \sin \frac{\pi x}{l} \times \frac{\pi}{l}$ $\frac{d^2y}{dx^2} = -\left[\frac{a_1\pi^2}{12}\sin\frac{\pi x}{1}\right] - \left[\frac{a_1\pi^2}{12}\sin\frac{\pi x}{1}\right]$

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bstitute Vale in equation (3)

$$U = \frac{EI}{2} \int_{0}^{1} \left[-\frac{a_{1}\pi^{2}}{l^{2}} Sin \frac{\pi x}{l} \right]^{2} dsc$$

$$U = \frac{EI}{2} \times \pi^{1} \int_{0}^{1} \left(a_{1}^{2} Sin^{2} \frac{\pi x}{l} \right) dsc - (4)$$

$$\int_{0}^{1} \left(a_{1}^{2} Sin^{2} \frac{\pi x}{l} \right) dx = a_{1}^{2} \int_{0}^{1} \frac{1}{2} \left(1 - \cos \frac{2\pi x}{l} \right) dsc - (4)$$

$$= \frac{a_{1}^{2}}{2} \int_{0}^{1} \left[1 - \cos \frac{4\pi x}{l} \right] dsc$$

$$= \frac{a_{1}^{2}}{2} \int_{0}^{1} \left[1 - \cos \frac{4\pi x}{l} \right] dsc$$

$$= \frac{a_{1}^{2}}{2} \left[(x)_{0}^{1} - \left[\frac{\sin \frac{4\pi x}{l}}{2\pi l} \right] dx \right]$$

$$= \frac{a_{1}^{2}}{2} \left[(x)_{0}^{1} - \left[\frac{\sin \frac{4\pi x}{l}}{2\pi l} \right] dx \right]$$

$$= \frac{a_{1}^{2}}{2} \left[1 - \frac{1}{2\pi} (s - s) \right] = \frac{a_{1}^{2} L}{2}$$

$$\int_{0}^{1} \left[\frac{a_{1}^{2} Sin^{2} \pi x}{l} \right] dx = \frac{a_{1}^{2} L}{2}$$

$$\int_{0}^{1} \left[\frac{a_{1}^{2} L}{2\pi l} \right] dx = \frac{a_{1}^{2} L}{2}$$

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$$\int_{0}^{1} \left[\frac{a_{1}^{2} L}{2\pi l} \right] dx = \frac{a_{1}^{2}$$





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Year & Branch :

III Aerospace Semester

Course : 19ASB302 – Finite Element Method for Aerospace

Prof & Head/ Aerospace

Unit: 1

We Know that, Work done, H = I wyda = (w (a2 sin TIX) dx by external force H = IN [[a, sin Tx] dx $= W \left[a_1 \left[-\cos \frac{\pi x}{J} \right]^2 \right]$ = W [-a, 1 [cos 12]] = W [-a, [-1-1]] = W [&ail] It = QWL [a,] --- (6) Substitute (5) and (6) values in equation (2) $=\frac{E\Gamma}{2}\times\frac{\pi^4}{13}\left(\frac{a_1^2L}{2}\right)-\frac{2wL}{\pi}\left(a_1\right)$ Π= ΕΙπ⁴ [a1²] - 2NJ [a1] For Stationary value of TT, The following conditions must be satisfied. STI = EITH 291 - 2WH = 0 [STI = 0]

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19ASB302 – Finite Element Method for Aerospace

Unit: 1

$$\frac{EI\pi^{4}}{41^{2}}2\alpha_{1} = \frac{2wI}{\pi}$$

$$\alpha_{1} = \frac{4wI^{4}}{EI\pi^{5}}$$

Substitute Value a, (1)

$$y = a_1 \sin \pi x$$

$$y = 4 \frac{4 \times 4}{5} \sin \pi x$$
EFT 15

are know that, maximum deflection occurs at oc= 42

Substituto x= 42 in equation

$$\frac{y_{\text{max}} = 4Wl^4}{EF\Pi^5} \sin \frac{\pi \times l_2}{l}$$

$$= \frac{4Wl^4}{EI\Pi^5} \sin \frac{\pi}{2} \quad \text{?' Sin } \frac{\pi}{2} = 1$$

we know that, simply supposted bearing Subjected to uniformly distributed load, ymax = 5 wet 384 EI

i.e, ymax = 0.01307 W14 Hence, exact solution and Raleigh-Ritz

Solution are compared.





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19ASB302 – Finite Element Method for Aerospace

Unit: 1

Bending moment at midspan

Bending moment, $M = EId^2y$ Que know equation (A) $\frac{d^2y}{dx^2} = -\left[\frac{a_1\pi^2}{1^2}\sin\frac{\pi_{22}}{e}\right]$ Substitute at value, $\frac{d^2y}{dx^2} = -\left[\frac{4\mu_0^4}{EI\pi^5} \times \frac{\pi^2}{1^2}\sin\frac{\pi_{22}}{e}\right]$ Maximum bending occurs at $\alpha = 1/2$ $\frac{d^2y}{dx^2} = -\left[\frac{4\mu_0^4}{EI\pi^5} \times \frac{\pi^2}{1^2}\sin\frac{\pi_1^4}{2}\right]$ $\frac{d^2y}{dx^2} = -\left[\frac{4\mu_0^4}{EI\pi^5} \times \frac{\pi^2}{1^2}\sin\frac{\pi_1^4}{2}\right]$ $\frac{d^2y}{dx^2} = -\left[\frac{4\mu_0^4}{EI\pi^5} \times \frac{\pi^2}{1^2}\sin\frac{\pi_1^4}{2}\right]$ $\frac{d^2y}{dx^2} = -\left[\frac{4\mu_0^4}{EI\pi^5} \times \frac{\pi^2}{1^2} \times 1\right]$ $\frac{d^2y}{dx^2} = -\left[\frac{4\mu_0^4}{EI\pi^5} \times \frac{\pi^2}{1^2} \times 1\right]$ $\frac{d^2y}{dx^2} = -\frac{4\mu_0^2}{EI\pi^3}$ $\frac{d^2y}{dx^2} = -\frac{4\mu_0^2}{EI\pi^3}$





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Unit: 1

Substitute dey value in bending moment equation

Mmid = EI x (-0.129) W12
EI

= -0.129w12 [Negative Sign indicates downword load]

rue know that, simply supposted beam maximum bending moment is

 $M_{mid} = \frac{wl^2}{3} = 0.125wl^2$

Hence, exact Solution and Rayleigh-Ritz Soluction are compared.

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering

